Thermodynamics

ME232

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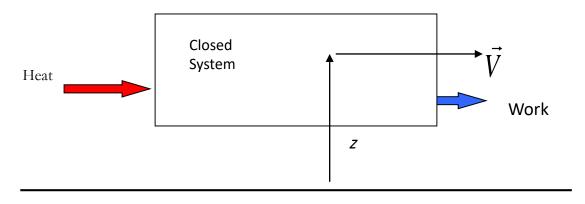
FIRST LAW OF THERMODYNAMICS

ENERGY ANALYSIS OF CLOSED SYSTEM

First Law of Thermodynamics

- The First Law is usually referred to as the Law of Conservation of Energy, i.e. *energy can neither be created nor destroyed, but rather transformed from one state to another.*
- The energy balance is maintained within the system being studied/defined boundary.
- The various energies associated are then being observed as they cross the boundaries of the system.

Energy Balance for Closed System



Reference Plane, z = 0

$$\begin{pmatrix}
\text{Total energy} \\
\text{entering the system}
\end{pmatrix} - \begin{pmatrix}
\text{Total energy} \\
\text{leaving the system}
\end{pmatrix} = \begin{pmatrix}
\text{The change in total} \\
\text{energy of the system}
\end{pmatrix}$$

or

$$E_{in} - E_{out} = \Delta E_{system}$$

☐ According to classical thermodynamics

$$Q_{net} - W_{net} = \Delta E_{system}$$

 \Box The total energy of the system, $E_{\rm system}$, is given as

$$E = Internal\ energy + Kinetic\ energy + Potential\ energy$$

$$E = U + KE + PE$$

☐ The change in stored energy for the system is

$$\Delta E = \Delta U + \Delta KE + \Delta PE$$

☐ The first law of thermodynamics for closed systems then can be written as

$$Q_{net} - W_{net} = \Delta U + \Delta KE + \Delta PE$$

☐ If the system does not move with a velocity and has no change in elevation, the conservation of energy equation is reduced to

$$Q_{net} - W_{net} = \Delta U$$

☐ The first law of thermodynamics can be in the form of

$$q_{net} - w_{net} = \left(u_2 - u_1 + \frac{V_2^2 - V_1^2}{2000} + \frac{g(z_2 - z_1)}{1000}\right) \qquad (kJ/kg)$$

$$Q_{net} - W_{net} = m \left(u_2 - u_1 + \frac{V_2^2 - V_1^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$
 (kJ)

☐ For a constant volume process,

$$Q_{net} - W_{net} = m \left(u_2 - u_1 + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

$$Q_{net} = m \left(u_2 - u_1 + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

☐ For a constant pressure process,

$$Q_{net} - W_{net} = m \left(u_2 - u_1 + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

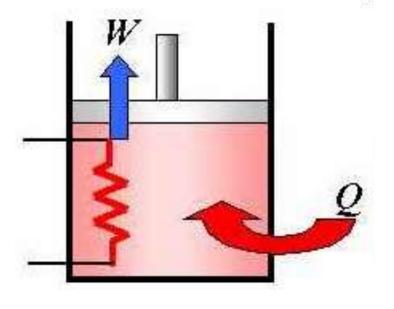
$$Q_{net} - P(V_2 - V_1) = m \left(u_2 - u_1 + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

$$Q_{net} = m \left(u_2 - u_1 + P(V_2 - V_1) + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

$$Q_{net} = m \left(h_2 - h_1 + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

Example of Closed Systems





Rigid tank

Piston cylinder

Example 3.1

A closed system of mass 2 kg undergoes an adiabatic process. The work done on the system is 30 kJ. The velocity of the system changes from 3 m/s to 15 m/s. During the process, the elevation of the system increases 45 meters. Determine the change in internal energy of the system.

* Rearrange the equation

$$Q_{net} - W_{net} = m \left(u_2 - u_1 + \frac{V_2^2 - V_1^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

$$-W_{net} = m \left(u_2 - u_1 + \frac{V_2^2 - V_1^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

$$-(-30) = 2\Delta u + 2 \left(\frac{15^2 - 3^2}{2000} \right) + 2 \left(\frac{9.81(45)}{1000} \right)$$

$$\Delta u = 14.451 \, kJ \, Ans..$$

Solution:

Energy balance,

$$Q_{net} - W_{net} = m \left(u_2 - u_1 + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$

Example 3.2

Steam at 1100 kPa and 92 percent quality is heated in a rigid container until the pressure is 2000 kPa. For a mass of 0.05 kg, calculate the amount of heat supply (in kJ) and the total entropy change (in kJ/kg.K).

Solution:

$$State1$$

$$at P_1 = 1100 kPa, x_1 = 0.92$$

$$v_1 = v_{f1} + x_1 v_{fg1}$$

$$= 0.00113 + 0.92(0.17753 - 0.001133)$$

$$= 0.1634 \frac{m^3}{kg}$$

$$u_1 = u_{f,1} + x_1 u_{fg,1}$$

$$= 779.78 + 0.92(1805.7)$$

$$= 2441.024 \frac{kJ}{kg}$$

$$s_1 = s_{f,1} + x_1 s_{fg,1}$$

$$= 2.1785 + 0.92(4.3735)$$

$$= 6.20212 \frac{m^3}{kg}$$

• For a rigid container, $v_2 = v_1 = 0.1634 \text{ m}^3/\text{kg}$

at
$$P_2 = 2000 \, kPa$$
, $v_2 = 0.1634 \, \frac{m^3}{kg}$

v	и	S
0.15122	2945.9	7.1292
0.1634	u_2	s_2
0.17568	3116.9	7.4337

$$u_2 = 2945.9 + \left(\frac{0.1634 - 0.15122}{0.17568 - 0.15122}\right) (3116.9 - 2945.9)$$
$$= 3030.42 \frac{kJ}{kg}$$

$$\begin{split} s_2 &= 7.1292 + \left(\frac{0.1634 - 0.15122}{0.17568 - 0.15122}\right) (7.4337 - 7.1292) \\ &= 7.2790 \frac{kJ}{kg.K} \end{split}$$

superheated

❖ Amount of heat supplied, Q

$$Q = m(u_2 - u_1)$$

$$= 0.05(3030.42 - 2441.9)$$

$$= 29.43 kJ$$

 \bullet The change in entropy, Δ s

$$\Delta s = s_2 - s_1$$
= 7.2790 - 6.204
= $\frac{1.075 \frac{kJ}{kg.K}}{\frac{kJ}{kg.K}}$

Example 3.3

A rigid tank is divided into two equal parts by a partition. Initially one side of the tank contains 5 kg water at 200 kPa and 25°C, and the other side is evacuated. The partition is then removed, and the water expands into the entire tank. The water is allowed to exchange heat with its surroundings until the temperature in the tank returns to the initial value of 25°C. Determine (a) the volume of the tank (b) the final pressure (c) the heat transfer for this process.

Solution:

State1

$$P_1 = 200 \, kPa$$
, $T_1 < T_{sat}$ Comp. liquid $T_1 = 25^{\circ} \, C$ $v_1 = v_{f@25^{\circ} \, C} = 0.001003 \frac{m^3}{kg}$

initial volume of half resevoir

$$V_1 = mv$$

$$= 5(0.001003)$$

$$\square 0.005 m^3$$

❖ The initial volume for entire tank

$$V_{resevoir} = 2(0.005)$$
$$= 0.01 m^3$$

❖ The final pressure

$$State 2$$

$$T_2 = 25^{\circ} C$$

$$v_f = 0.001003 \frac{m^3}{kg}$$

$$v_2 = \frac{0.01}{5} = 0.002 \frac{m^3}{kg}$$

$$v_g = 43.34 \frac{m^3}{kg}$$

$$check \ region!$$

$$v_f < v < v_g \rightarrow saturated \ mixture$$

$$then : P_2 = P_{sat} = 3.169 \ kPa$$

❖ The heat transfer for this process

$$Q_{net} - W_{net} = m(\Delta u + \Delta ke + \Delta Pe)$$
 $Q_{net} - W_{net} = m(\Delta u + \Delta ke + \Delta Pe)$
 $Q_{net} = m\Delta u = m(u_2 - u_1)$

$$\Rightarrow u_{1} = u_{f @ 25^{\circ}C} = 104.83 \frac{kJ}{kg}$$

$$u_{2} = u_{f} + x_{2}u_{fg}$$

$$x_{2} = \frac{v_{2} - v_{f}}{v_{fg}}$$

$$= 2.3 \times 10^{-5}$$

$$\therefore u_{2} = 104.83 + 2.3 \times 10^{-5} \quad (2304.3)$$

$$= 104.88 \frac{kJ}{kg}$$
Then:
$$Q_{net} = 5 \quad (104.88-104.83)$$

$$= 0.25 \quad kJ$$

♦ +ve sign indicates heat transfer into the system.

Supplementary Problems 1

- 1. Two tanks are connected by a valve. Tank A contains 2 kg of carbon monoxide gas at 77°C and 0.7 bar. Tank B holds 8 kg of the same gas at 27°C and 1.2 bar. Then the valve is opened and the gases are allowed to mix while receiving energy via heat transfer from the surrounding. The final equilibrium temperature is found to be 42°C. Determine (a) the final pressure (b) the amount of heat transfer. Also state your assumption. $[P_2=105 \text{ kPa}, Q=+37.25 \text{ kJ}]$
- 2. A piston cylinder device contains 0.2 kg of water initially at 800 kPa and 0.06 m³. Now 200 kJ of heat is transferred to the water while its pressure is held constant. Determine the final temperature of the water. Also, show the process on a T-V diagram with respect to saturation lines. [721.1°C]

Supplementary Problems 1

3. A piston-cylinder device contains 6 kg of refrigerant-134a at 800 kPa and 50°C. The refrigerant is now cooled at constant pressure until it exist as a liquid at 24°C. Show the process on T-v diagram and determine the heat loss from the system. State any assumption made.

 $[1210.26 \, kJ]$

4. A 0.5 m³ rigid tank contains refrigerant-134a initially at 200 kPa and 40 percent quality. Heat is now transferred to the refrigerant until the pressure reaches 800 kPa. Determine (a) the mass of the refrigerant in the tank and (b) the amount of heat transferred. Also, show the process on a P-v diagram with respect to saturation lines.

[12.3 kg, 2956.2

5. An insulated tank is divided into two parts by a partition. One part of the tank contains 6 kg of an ideal gas at 50°C and 800 kPa while the other part is evacuated. The partition is now removed, and the gas expands to fill the entire tank. Determine the final temperature and the pressure in the tank.

[50°C, 400 kPa]

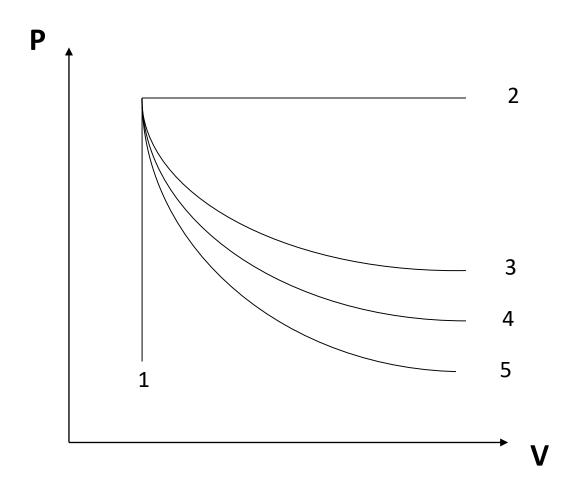
Closed System First Law of a Cycle

- Some thermodynamic cycle composes of processes in which the working fluid undergoes a series of state changes such that the final and initial states are identical.
- ❖ For such system the change in internal energy of the working fluid is zero.
- ❖ The first law for a closed system operating in a thermodynamic cycle becomes

$$Q_{net} - W_{net} = \Delta U_{cycle}$$

$$Q_{net} = W_{net}$$

Boundary Works



According to a law of $PV^n = \text{constant}$

No	Value of n	Process	Description	Result of IGL
1	∞	iso <i>cho</i> ric	constant volume $(V_1 = V_2)$	$\frac{P_1}{T_1} = \frac{P_2}{T_2}$
2	0	iso <i>bar</i> ic	constant pressure $(P_1 = P_2)$	$\frac{V_1}{T_1} = \frac{V_2}{T_2}$
3	1	iso <i>thermal</i>	constant temperature $(T_1 = T_2)$	$P_1V_1=P_2V_2$
4	1 <n< td="" γ<=""><td>polytropic</td><td>-none-</td><td>$\frac{P_1}{P_2} = \left(\frac{V_2}{V_1}\right)^n = \left(\frac{T_1}{T_2}\right)^{\frac{n}{n-1}}$</td></n<>	polytropic	-none-	$\frac{P_1}{P_2} = \left(\frac{V_2}{V_1}\right)^n = \left(\frac{T_1}{T_2}\right)^{\frac{n}{n-1}}$
5	γ	i <i>sentropi</i> c	constant entropy $(S_1 = S_2)$	$P_2 V_1 T_2$

☐ Various forms of work are expressed as follows

Process	Boundary Work		
iso <i>cho</i> ric	$W_{12} = P(V_2 - V_1) = 0$		
iso <i>bar</i> ic	$W_{12} = P(V_2 - V_1)$		
iso <i>thermal</i>	$W_{12} = P_1 V_1 \ln \frac{V_2}{V_1}$		
polytropic	$W_{12} = \frac{P_2 V_2 - P_1 V_1}{1 - n}$		
i <i>sentropi</i> c			

Example 3.4

Sketch a P-V diagram showing the following processes in a cycle

- **Process 1-2:** isobaric work output of 10.5 kJ from an initial volume of 0.028 m³ and pressure 1.4 bar,
- **Process 2-3**: isothermal compression, and
- **Process 3-1**: isochoric heat transfer to its original volume of 0.028 m³ and

pressure 1.4 bar.

Calculate (a) the maximum volume in the cycle, in m³, (b) the isothermal work, in kJ, (c) the net work, in kJ, and (d) the heat transfer during isobaric expansion, in kJ.

Solution:

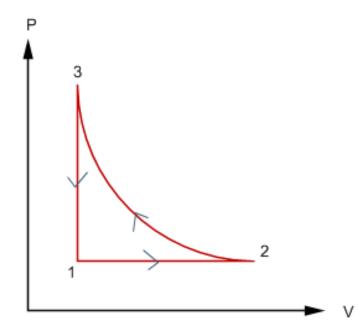
Process by process analysis,

Section
$$1-2(isobaric)$$

$$W_{12} = P(V_2 - V_1) = 10.5$$

$$140(V_2 - 0.028) = 10.5$$

$$V_2 = 0.103 \, m^3$$



❖ The isothermal work

Section 2-3 (isothermal)

$$P_2V_2 = P_3V_3$$

$$P_3 = \left(\frac{0.103}{0.028}\right) (140) = 515 \, kPa$$

$$\rightarrow W_{23} = P_2 V_2 \ln \frac{V_3}{V_2}$$
$$= 140(0.103) \ln \left(\frac{0.028}{0.103}\right)$$

$$= -18.78 kJ$$

❖ The net work

$$W_{31} = 0$$

$$\therefore W_{net} = W_{12} + W_{23} + W_{31}$$

$$= 10.5 - 18.78$$

$$= -8.28 kJ$$

Example 3.5

A fluid at 4.15 bar is expanded reversibly according to a law PV = constant to a pressure of 1.15 bar until it has a specific volume of $0.12 \text{ m}^3/\text{kg}$. It is then cooled reversibly at a constant pressure, then is cooled at constant volume until the pressure is 0.62 bar; and is then allowed to compress reversibly according to a law PV^n = constant back to the initial conditions. The work done in the constant pressure is 0.525 kJ, and the mass of fluid present is 0.22 kg. Calculate the value of n in the fourth process, the net work of the cycle and sketch the cycle on a P-V diagram.

Solution:

Process by process analysis,

Section
$$1-2$$
 (isothermal)

$$P_1V_1 = P_2V_2$$

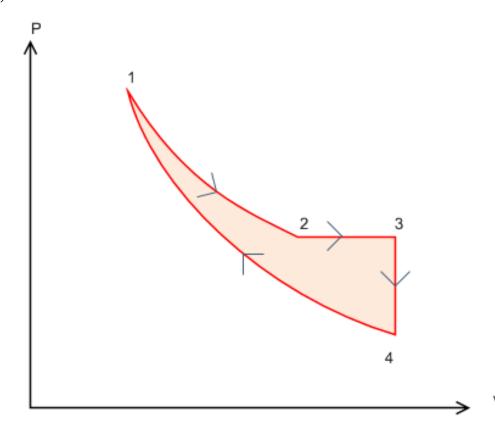
$$V_1 = \left(\frac{115}{415}\right)0.22(0.12)$$

$$= 0.00732 \, m^3$$

$$W_{12} = P_1 V_1 \ln \frac{V_2}{V_1}$$

$$= 415 (0.00732) \ln \frac{0.0264}{0.00732}$$

$$= 3.895 kJ$$



Section 2-3 (isobaric)

$$W_{23} = P(V_3 - V_2) = 0.525 \, kJ$$

$$V_3 = \frac{0.525}{115} + 0.0264$$

$$= 0.03097 \, m^3$$

Section 3-4 (isochoric)

$$W_{34} = 0$$

Section 4-1(PolytroPic)

$$\frac{P_4}{P_1} = \left(\frac{V_1}{V_4}\right)^n$$

$$\frac{62}{415} = \left(\frac{0.00732}{0.03097}\right)^n$$

$$\ln 0.1494 = n \ln 0.2364$$

$$n = \underline{1.3182}$$

$$W_{41} = \frac{P_1 V_1 - P_4 V_4}{1 - n}$$

$$= \frac{415(0.0072) - 62(0.03097)}{1 - 1.3182}$$

$$= -3.5124 \, kJ$$

The net work of the cycle

$$W_{net} = W_{12} + W_{23} + W_{34} + W_{41}$$
$$= 0.9076 \, kJ$$

Supplementary Problems 2

1. A mass of 0.15 kg of air is initially exists at 2 MPa and 350°C. The air is first expanded isothermally to 500 kPa, then compressed polytropically with a polytropic exponent of 1.2 to the initial state. Determine the boundary work for each process and the net work of the cycle.

2. 0.078 kg of a carbon monoxide initially exists at 130 kPa and 120°C. The gas is then expanded polytropically to a state of 100 kPa and 100°C. Sketch the P-V diagram for this process. Also determine the value of n (index) and the boundary work done during this process.

[1.248,1.855 kJ]

3. Two kg of air experiences the three-process cycle shown in Fig. 3-14. Calculate the net work.

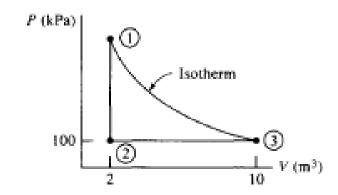


Fig. 3-14

4. A system contains 0.15 m³ of air pressure of 3.8 bars and 150° C. It is expanded adiabatically till the pressure falls to 1.0 bar. The air is then heated at a constant pressure till its enthalpy increases by 70 kJ. Sketch the process on a P-V diagram and determine the total work done.

Use $c_p=1.005$ kJ/kg.K and $c_v=0.714$ kJ/kg.K

FIRST LAW OF THERMODYNAMICS

MASS & ENERGY ANALYSIS OF CONTROL VOLUME

Conservation of Mass

- Conservation of mass is one of the most fundamental principles in nature. We are all familiar with this principle, and it is not difficult to understand it!
- For closed system, the conservation of mass principle is implicitly used since the mass of the system remain constant during a process.
- ➤ However, for control volume, mass can cross the boundaries. So the amount of mass entering and leaving the control volume must be considered.

Mass and Volume Flow Rates

Mass flow through a cross-sectional area per unit time is called the mass flow rate. Note the dot over the mass symbol indicates a time rate of change. It is expressed as

$$\dot{m} = \int \rho V . dA$$

➤ If the fluid density and velocity are constant over the flow cross-sectional area, the mass flow rate is

$$\dot{m} = \rho AV = \frac{AV}{v}$$

$$where v = \frac{1}{\rho}$$

v iscalled specific voulme

Principal of Conservation of Mass

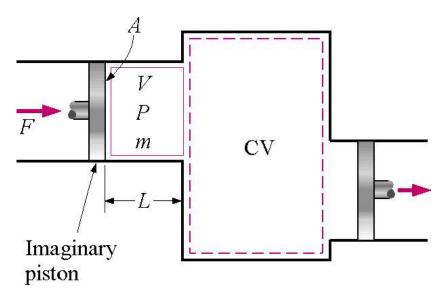
The conservation of mass principle for a control volume can be expressed as

$$\dot{m}_{in} - \dot{m}_{out} = \dot{m}_{CV}$$

For a steady state, steady flow process the conservation of mass principle becomes

$$\dot{m}_{in} = \dot{m}_{out}$$
 (kg/s)

Flow Work & The Energy of a Flowing Fluid



As the fluid upstream pushes mass across the control volume, work done on that unit of mass is

$$\delta W_{flow} = F dL = F dL \frac{A}{A} = P dV = P v \delta m$$
$$\delta W_{flow} = \frac{\delta W_{flow}}{\delta m} = P v$$

Total Energy of a Flowing Fluid

The total energy carried by a unit of mass as it crosses the control surface is the sum of the internal energy + flow work + potential energy + kinetic energy

$$\sum energy = u + Pv + \frac{V^{2}}{2} + gz = h + \frac{V^{2}}{2} + gz$$

> The first law for a control volume can be written as

$$\dot{Q}_{net} - \dot{W}_{net} = \sum_{out} \dot{m}_{out} \left(h_{out} + \frac{V_{out}^{2}}{2} + gz_{out} \right) - \sum_{in} \dot{m}_{in} \left(h_{in} + \frac{V_{in}^{2}}{2} + gz_{in} \right)$$

Total Energy of a Flowing Fluid

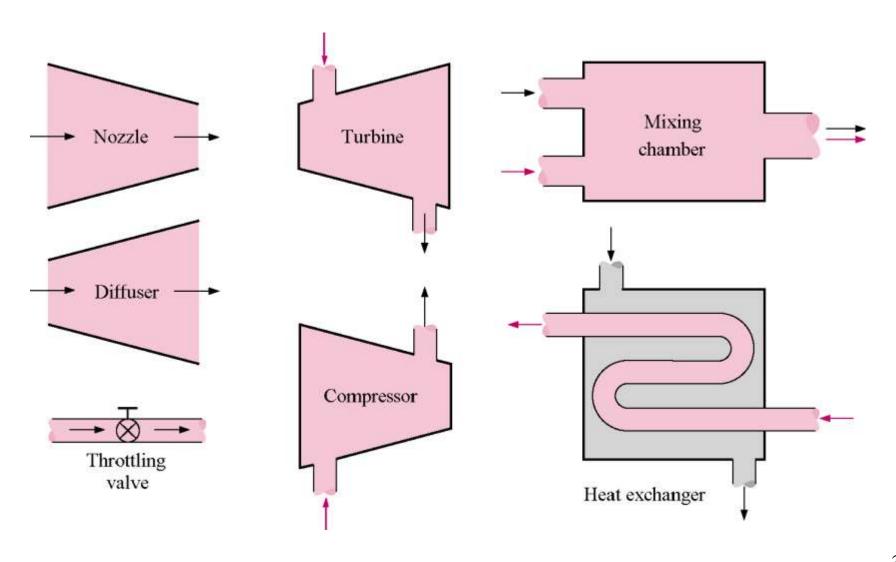
The steady state, steady flow conservation of mass and first law of thermodynamics can be expressed in the following forms

$$q_{net} - w_{net} = \left(h_2 - h_1 + \frac{V_2^2 - V_1^2}{2000} + \frac{g(z_2 - z_1)}{1000}\right) \qquad (kJ/kg)$$

$$Q_{net} - W_{net} = m \left(h_2 - h_1 + \frac{V_2^2 - V_1^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$
 (kJ)

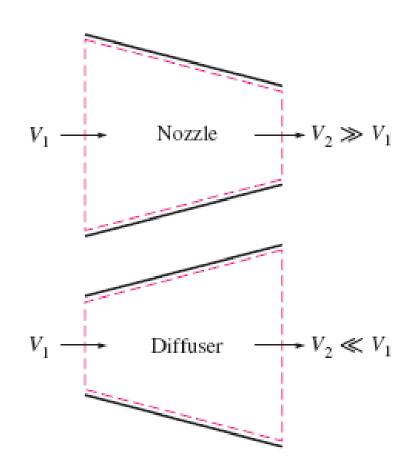
$$\dot{Q}_{net} - \dot{W}_{net} = \dot{m} \left(h_2 - h_1 + \frac{{V_2}^2 - {V_1}^2}{2000} + \frac{g(z_2 - z_1)}{1000} \right)$$
 (kW)

Steady-flow Engineering Devices



Nozzle & Diffuser

- Nozzle device that increases the velocity fluid at the expense of pressure.
- ☐ Diffuser device that increases pressure of a fluid by slowing it down.
- ☐ Commonly utilized in jet engines, rockets, space-craft and even garden hoses.
- $\mathbf{Q} = 0$ (heat transfer from the fluid to surroundings very small
- \Box W = 0 and \triangle PE = 0



☐ Energy balance (nozzle & diffuser):

$$\dot{Q}_{in} + \dot{W}_{in} + \sum_{in} \dot{m}_{in} \left(h_{in} + \frac{V_{in}^{2}}{2} + gz_{in} \right) = \dot{Q}_{out} + \dot{W}_{out} + \sum_{out} \dot{m}_{out} \left(h_{out} + \frac{V_{out}^{2}}{2} + gz_{out} \right)$$

$$m_{in} \left(h_{in} + \frac{V_{in}^{2}}{2} \right) = m_{out} \left(h_{out} + \frac{V_{out}^{2}}{2} \right)$$

$$\left(h_1 + \frac{V_1^2}{2}\right) = \left(h_2 + \frac{V_2^2}{2}\right)$$

Steam at 0.4 MPa, 300°C, enters an adiabatic nozzle with a low velocity and leaves at 0.2 MPa with a quality of 90%. Find the exit velocity.

Solution:

State 1State 2 $P_1 = 0.4 MPa$ $P_2 = 0.2 MPa$ $T_1 = 300^{\circ} C$ $x_2 = 0.9$ $V_1 \square 0$

Simplified energy balance:

$$\begin{pmatrix} h_1 + \frac{y_1^2}{2} \\ h_2 + \frac{V_2^2}{2} \end{pmatrix} = \begin{pmatrix} h_2 + \frac{V_2^2}{2} \\ h_2 + \frac{V_2^2}{2} \end{pmatrix}$$

$$\begin{bmatrix} State 1 \\ P_1 = 0.4 \, MPa \\ h_1 = 3067.1 \frac{kJ}{kg} \\ T_1 = 300^{\circ} \, C \\ \end{bmatrix} \text{ (sup } erheated \text{)}$$

$$\begin{bmatrix} State 2 \\ P_2 = 0.2 \, MPa \\ h_2 = h_f + x_2 h_{fg} \\ x_2 = 0.9 \\ \end{bmatrix} h_2 = 2486.1 \frac{kJ}{kg}$$

***** Exit velocity:

$$V_2 = \sqrt{2000(3067.1 - 2486.1)}$$
$$= 1078 \, m / s$$

Air at 10°C and 80 kPa enters the *diffuser* of a jet engine steadily with a velocity of 200 m/s. The inlet area of the diffuser is 0.4 m². The air leaves the diffuser with a velocity that is very small compared with the inlet velocity. Determine (a) the mass flow rate of the air and (b) the temperature of the air leaving the diffuser.

State 1

State 2

$$P_{1} = 80 kPa \qquad V_{2} \square 0$$

$$T_{1} = 10^{\circ} C$$

$$V_{1} = 200 m / s$$

$$A_{1} = 0.4 m^{2}$$

Solution:

Simplified energy balance:

$$\left(h_{1} + \frac{V_{1}^{2}}{2}\right) = \left(h_{2} + \frac{V_{2}^{2}}{2}\right)$$

From Ideal Gas Law:

$$v_1 = \frac{RT_1}{P_1} = 1.015 \frac{m^3}{kg}$$

Mass flow rate

$$\dot{m} = \frac{1}{v_1} V_1 A_1$$

$$= \left(\frac{1}{1.015}\right) (200)(0.4)$$

$$= \frac{78.8 \frac{kg}{s}}{\frac{s}{s}}$$

Enthalpy at state 1

$$h_1 = C_p T_1 = 1.005 (283)$$
$$= 284.42 \frac{kJ}{kg}$$

From energy balance:

$$h_2 = h_1 + \frac{V_1^2}{2000}$$

$$= 284.42 + \frac{200^2}{2000}$$

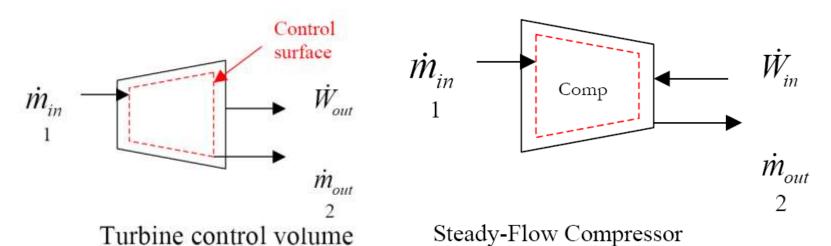
$$= 304.42 \frac{kJ}{kg}$$

$$T_2 = \frac{h_2}{C_p}$$

$$= \frac{304.42}{1.005}$$

$$= 302.9 K$$

Turbine & Compressor



- ☐ Turbine a work producing device through the expansion of a fluid.
- ☐ Compressor (as well as pump and fan) device used to increase pressure of a fluid and involves work input.
- $\mathbf{Q} = 0$ (well insulated), $\Delta PE = 0$, $\Delta KE = 0$ (very small compare to Δ enthalpy).

☐ Energy balance: for turbine

$$\dot{Q}_{in} + \dot{W}_{in} + \sum_{in} \dot{m}_{in} \left(h_{in} + \frac{V_{in}^{2}}{2} + gz_{in} \right) = \dot{Q}_{out} + \dot{W}_{out} + \sum_{out} \dot{m}_{out} \left(h_{out} + \frac{V_{out}^{2}}{2} + gz_{out} \right)$$

$$\dot{m}_{in}(h_{in}) = \dot{W}_{out} + \dot{m}_{out}(h_{out})$$

$$\dot{W}_{out} = \dot{m} (h_1 - h_2)$$

Energy balance: for compressor, pump and fan

$$\dot{Q}_{in} + \dot{W}_{in} + \sum_{in} \dot{m}_{in} \left(h_{in} + \frac{V_{in}^{2}}{2} + gz_{in} \right) = \dot{Q}_{out} + \dot{W}_{out} + \sum_{out} \dot{m}_{out} \left(h_{out} + \frac{V_{out}^{2}}{2} + gz_{out} \right)$$

$$\dot{W}_{in} + \dot{m}_{in}(h_{in}) = \dot{m}_{out}(h_{out})$$

$$\dot{W}_{in} = \dot{m} \left(h_2 - h_1 \right)$$

The power output of an adiabatic steam turbine is 5 MW. Compare the magnitudes of Δh , Δke , and Δpe . Then determine the work done per unit mass of the steam flowing through the turbine and calculate the mass flow rate of the steam.

Data : Inlet (P = 2 MPa, T =
$$400^{\circ}$$
C,v = 50 m/s, z = 10 m)
Exit (P = 15 kPa, x = 90% , v = 180 m/s, z = 6 m)

Solution:

State 1

$$p_1 = 2MPa$$
 sup erheated
 $T_1 = 400^{\circ} C$ $h_1 = 3248.4 \frac{kJ}{kg}$

State 2

$$P_2 = 15 kPa$$

$$x_2 = 0.9$$
 sat.mixture

$$h_2 = h_{f2} + x_2 h_{fg2}$$

$$= 225.94 + 0.9 (2372.3)$$

$$= 2361.01 \frac{kJ}{kg}$$

From energy balance:

$$\dot{Q}_{in} + \dot{W}_{in} + \sum_{in} \dot{m}_{in} \left(h_{in} + \frac{V_{in}^{2}}{2} + gz_{in} \right) =
\dot{Q}_{out} + \dot{W}_{out} + \sum_{out} \dot{m}_{out} \left(h_{out} + \frac{V_{out}^{2}}{2} + gz_{out} \right)$$

Solve the equation:

$$\Delta h = h_2 - h_1 = -887.39 \frac{kJ}{kg}$$

$$\Delta KE = \frac{V_2^2 - V_1^2}{2000} = 14.95 \frac{kJ}{kg}$$

$$\Delta PE = \frac{g(z_2 - z_1)}{1000} = -0.04 \frac{kJ}{kg}$$

the work done per unit mass

$$W_{out} = \left[(h_1 - h_2) + \left(\frac{V_1^2 - V_2^2}{2000} \right) + \left(\frac{g(z_1 - z_2)}{1000} \right) \right]$$

$$= 887.39 - 14.95 + 0.04$$

$$= 872.48 \frac{kJ}{kg}$$

The mass flow rate

$$\dot{m} = \frac{\dot{W}_{out}}{W_{out}} = \frac{5000}{872.48} = \frac{5.73 \frac{kg}{s}}{872.48}$$

Air at 100 kPa and 280 K is compressed steadily to 600 kPa and 400 K. The mass flow rate of the air is 0.02 kg/s, and a heat loss of 16 kJ/kg occurs during the process. Assuming the changes in kinetic and potential energies are negligible, determine the necessary power input to the compressor.

Solution:

* simplified energy balance:

$$\dot{W}_{in} = \dot{m}(h_2 - h_1) + \dot{Q}_{out}$$
$$= \dot{m}(h_2 - h_1) + \dot{m}q_{out}$$

$$State 1$$

$$P_{1} = 100 \, kPa \qquad air$$

$$T_{1} = 280 \, K \quad h_{1} = 280.13 \frac{kJ}{kg}$$

$$State 2$$

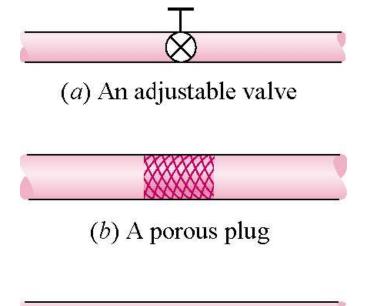
$$P_{2} = 600 \, kPa \qquad air$$

$$T_{2} = 400 \, K \quad h_{2} = 400.98 \frac{kJ}{kg}$$

Thus

$$\dot{W}_{in} = 0.02 \left[(400.98 - 280.13) + 16 \right]$$
$$= 2.74 \, kW$$

Throttling Valve



(c) A capillary tube

Flow-restricting devices that cause a significant pressure drop in the fluid.

Some familiar examples are ordinary adjustable valves and capillary tubes.

State 1

$$P_1 = 8000 \, kPa$$
 sup erheated
 $T_1 = 300^{\circ} \, C$ $h_1 = 2786.5 \frac{kJ}{kg}$

State 2

$$P_2 = 1600 \, kPa$$

$$h_2 = h_1$$
make interpolation

Steam enters a throttling valve at 8000 kPa and 300°C and leaves at a pressure of 1600 kPa. Determine the final temperature and specific volume of the steam.

P(kPa)	$T({}^{o}C)$	v_f	v_g	h_{f}	h_{g}
1500	198.29	0.001154	0.131710	844.55	2791
1600	T_2	v_{f2}	v_{g2}	$h_{\!f2}$	h_{g2}
1750	205.72	0.001166	0.113440	878.16	2795.2

☐ At state 2, the region is sat. ☐ mixture

$$T_2 = T_{sat} = \underline{201.3^{\circ} C}$$

☐ Getting the quality at state 2

$$x_2 = \frac{h_2 - h_{f2}}{h_{g2} - h_{f2}}$$

$$= \frac{2786.5 - 857.994}{2792.68 - 857.994}$$

$$= 0.997$$

☐ Specific volume at state 2

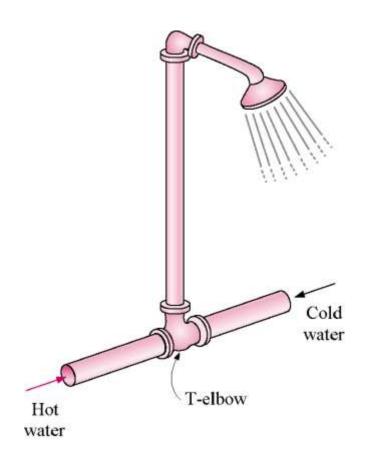
$$v_2 = v_{f2} + x_2 v_{fg2}$$

$$= 0.0011588 +$$

$$0.997 (0.124402 - 0.0011588)$$

$$= 0.1240 \frac{m^3}{kg}$$

Mixing Chamber



The section where the mixing process takes place.

An ordinary T-elbow or a Y-elbow in a shower, for example, serves as the mixing chamber for the cold- and hot-water streams.

Mixing Chamber

☐ Energy Balance:

$$\dot{m}_1 h_1 + \dot{m}_2 h_2 = \dot{m}_3 h_3$$

$$\dot{m}_1 h_1 + (\dot{m}_3 - \dot{m}_1) h_2 = \dot{m}_3 h_3$$

$$\dot{m}_1 (h_1 - h_2) = \dot{m}_3 (h_3 - h_2)$$

$$\dot{m}_1 = \dot{m}_3 \left(\frac{h_3 - h_2}{h_1 - h_2} \right)$$

the minimum mass flux of the water so that the water does not completely vaporize

$$\dot{m}_{w} = \frac{m_{s}C_{p,s} (T_{1s} - T_{2s})}{(h_{2w} - h_{1w})}$$

$$= \frac{100(1.25)(450 - 350)}{2794.2 - 88.61}$$

$$= 4.62 \frac{kg}{s}$$

☐ the rate of heat transfer

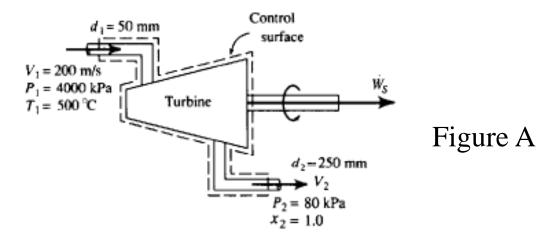
$$\dot{Q}_{w} = \dot{m}_{w} (h_{2w} - h_{1w})$$

$$= 4.62 (2794.2 - 88.61)$$

$$= 12.5 MW$$

Supplementary Problems 3

- 1. Air flows through the supersonic nozzle. The inlet conditions are 7 kPa and 420°C. The nozzle exit diameter is adjusted such that the exiting velocity is 700 m/s. Calculate (a) the exit temperature, (b) the mass flux, and (c) the exit diameter. Assume an adiabatic quasiequilibrium flow.
- 2. Steam at 5 MPa and 400°C enters a nozzle steadily velocity of 80 m/s, and it leaves at 2 MPa and 300°C. The inlet area of the nozzle is 50 cm², and heat is being lost at a rate of 120 kJ/s. Determine (a) the mass flow rate of the steam, (b) the exit velocity of the steam, and (c) the exit area nozzle.
- 3. Steam enters a turbine at 4000 kPa and 500°C and leaves as shown in Fig A below. For an inlet velocity of 200 m/s, calculate the turbine power output. (a) Neglect any heat transfer and kinetic energy change (b) Show that the kinetic energy change is negligible.



4. Consider an ordinary shower where hot water at 60°C is mixed with cold water at 10°C. If it is desired that a steady stream of warm water at 45°C be supplied, determine the ratio of the mass flow rates of the hot to cold water. Assume the heat losses from the mixing chamber to be negligible and the mixing to take place at a pressure of 150 kPa.